

# *SPS : Solar Power Satellite*



*July 2021*

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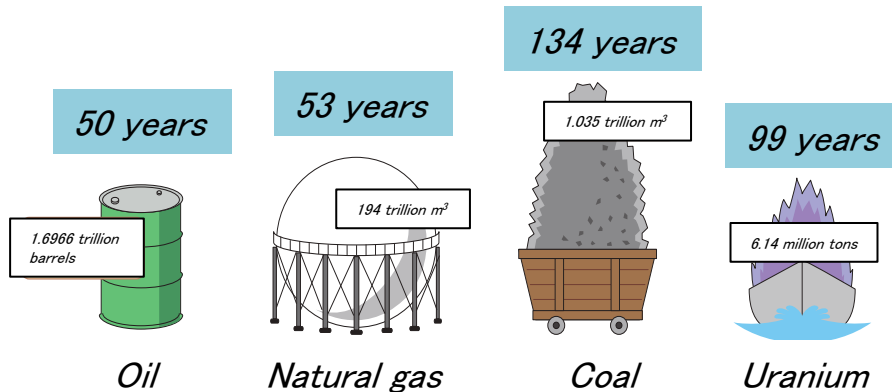


***1. Global energy problems and environmental problems***

# Energy problems and global environmental problems

## Limited Energy Resources

### Proven Reserves

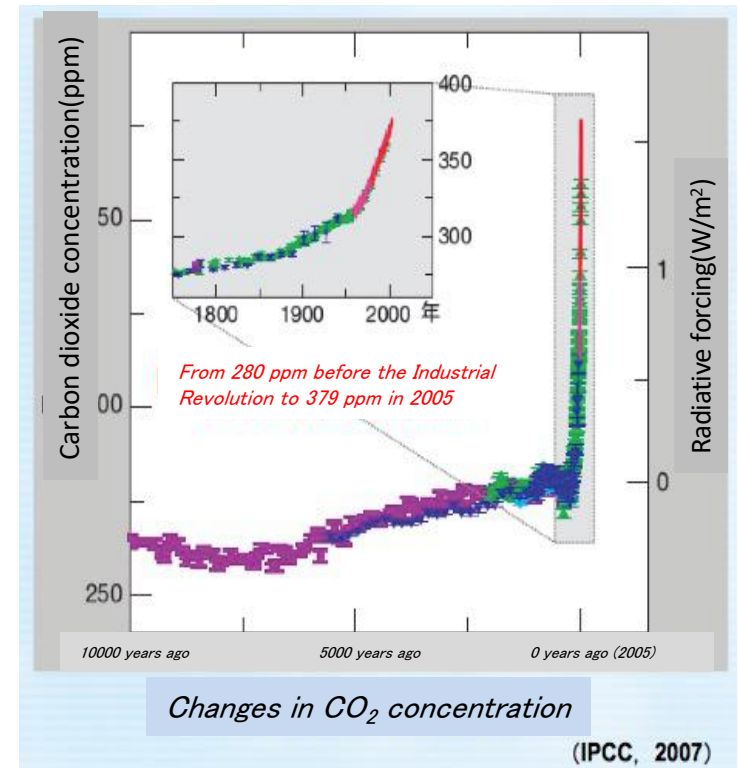


Reference : Japan Atomic Energy Relations Organization

### Consumption of fossil fuel

The earth accumulated solar energy in the form of fossil fuel over hundreds of millions of years. Mankind is using it up in just 100–150 years.

Energy acquisition by burning oil, coal, natural gas, etc. brings about deterioration of the global environment.

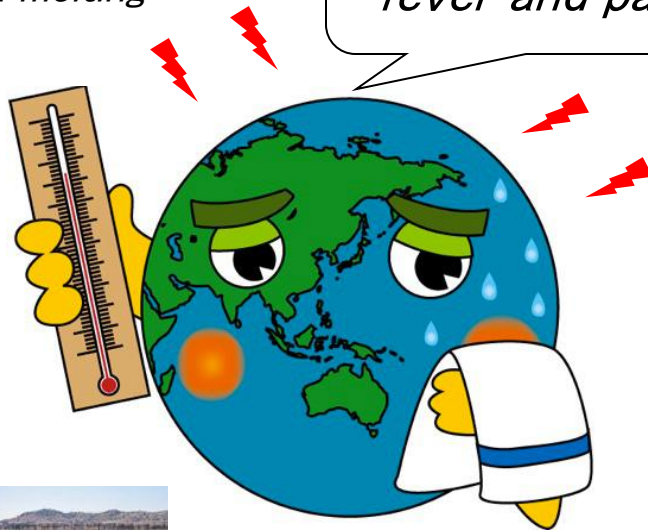


► We need a new clean energy system.

# *Is human being a virus for the Earth?*



*Glacial melting*



*Erath*



*Desertification*



*Other Planet*

► *To avoid this situation, ...*

*It is difficult to solve global problems within the closed system of the earth. We have to look for a solution outside the earth, i.e., in space.*

- *In space, unlike on earth, there is a vast field and plenty of solar energy that is not affected by weather.*
- *The concept of Solar Power Satellite (SPS) is to use space, the frontier of humanity, as a place for human to obtain energy. SPS has great potential as a clean and large-scale energy system.*
- *If the energy that supports modern society can be obtained in sufficient quantities by an environmentally friendly system, the serious problems that confront modern society will be solved and mankind will be able to take a new step toward further development.*
- *At this stage, SPS has not been shown as the best option for the future energy system, but it is a viable option as the technical principles for SPS have been already verified.*
- *SPS that collects natural solar energy on a large scale has an intrinsic advantage in terms of safety in case of accidents compared to the power systems that artificially generate large amounts of energy in a spatially limited area.*
- *Basic research for the realization of SPS has been progressing rapidly in recent years, and the level of technologies has already reached a level where full-scale experiments of SPS can be started to verify its feasibility as an energy system.*

► *We are at the stage where we should start full-scale space demonstration experiments to show that SPS can truly be the savior of human society.*

## ***2. Concepts and features***

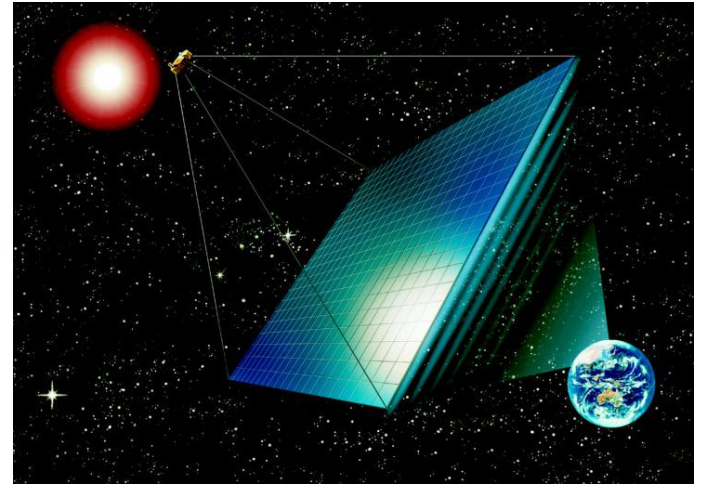


# *What is the concept of SPS ?*

*Transportation of the parts to space using reusable vehicles*



*SPS construction by assembling the parts*



*Solar power plant on the ground*

- ▶ *Simple and clear concept*
- *Transport the solar panels and microwave transmitters to space*
- *Assemble the solar power plant and microwave transmission system in space*
- *Transmit the generated power to the ground by microwave*



# Why do we think of Solar Power Satellite?

## Why the Sun?

Power from the sun to the earth is  $1.77 \times 10^{17}$  Watt, that is 10,000 times more than global power consumption.

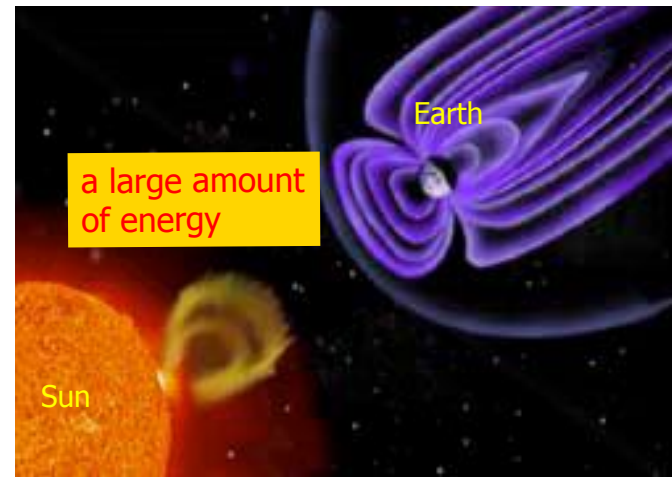
- ▶ Solar energy has great potential as an energy source for humankind.

## Why outer space?

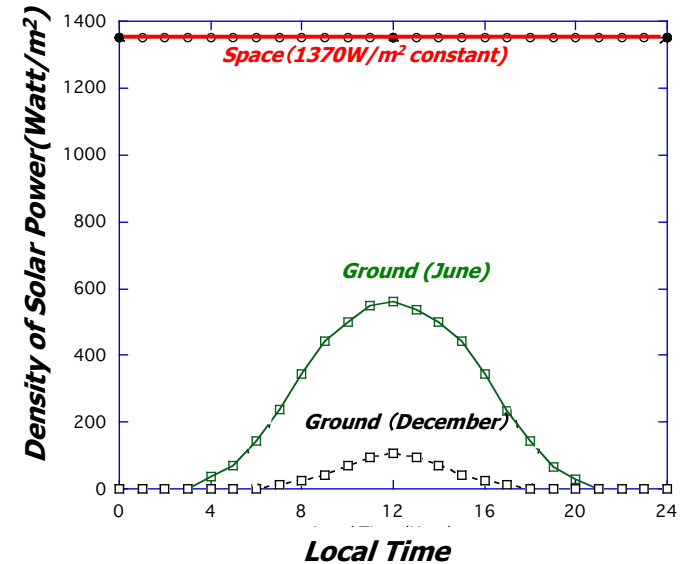
Power density of solar energy in space near the earth is  $1,370 \text{ W/m}^2$ .

Power density of solar energy on the ground is reduced to  $100 \sim 200 \text{ W/m}^2$  on average, due to the day/night cycle and weather/atmospheric effects.

- ▶ If efficient power transmission from outer space to the ground is possible, it is advantageous to acquire solar energy in space.

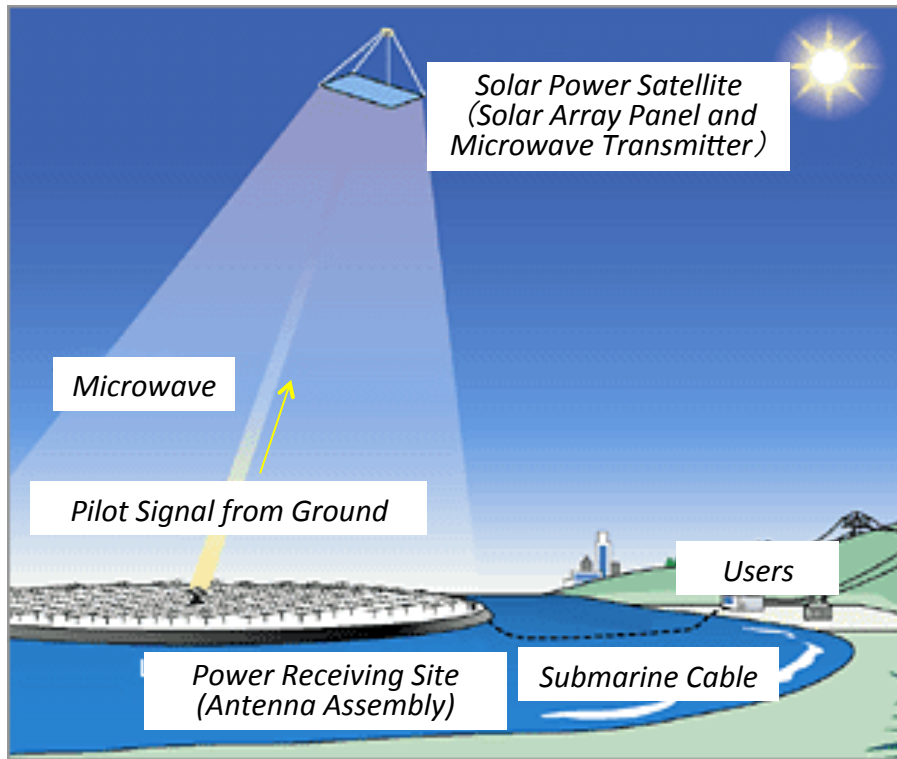


600 million tons of hydrogen burns per second (nuclear fusion) in the sun.



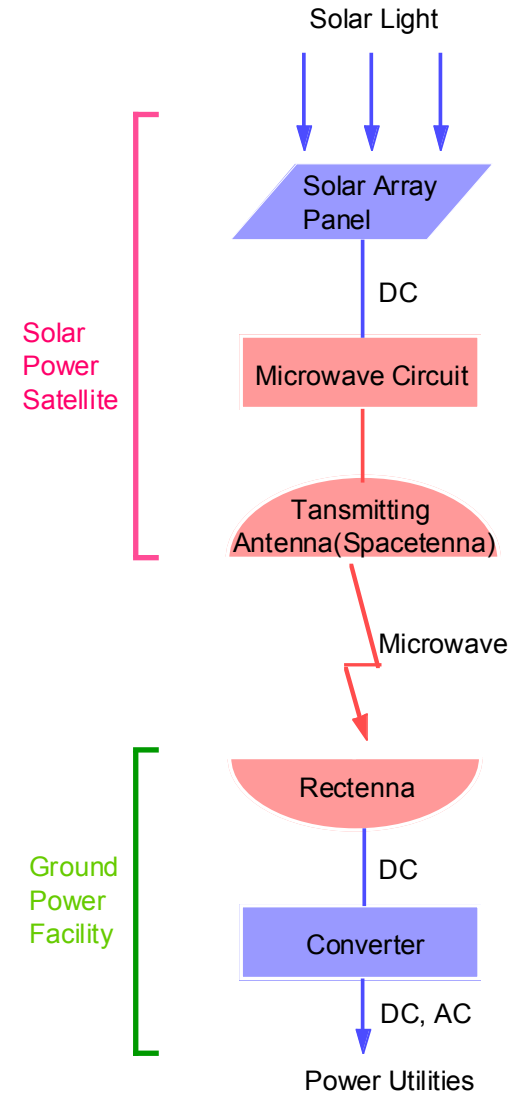
Density of solar power on ground and in space

# Configuration of Solar Power Satellite System



► SPS converts solar energy into electric power in space and transmits the power using microwave to the ground. On the ground, the received microwave is converted to commercial power and distributed to users through the existing power grid.

Space Solar Power Systems



# *Advantages of Solar Power Satellite System*

*Solar energy can be collected more and stably in space.*

## **Solar Power Plant on Ground**



***Rainy and cloudy days***



***Night***

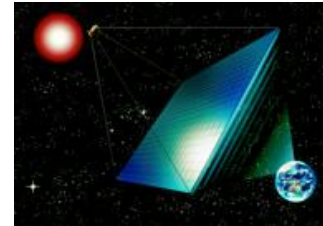


***Land problem***

## **Solar Power Plant in Space**



***No rainy or cloudy day***



***No night (depending on the orbit)***



***No land problem***

**Contrast**  
↔

# *Can Solar Power Satellite System get enough energy?*

## *–Example of energy flow model–*

<i>Process</i>	<i>NASA Reference System</i>	<i>USEF Tether System</i>	<i>Efficiency %</i>
			<i>JAXA Reference Model (2004)</i>
<i>Light collection</i>	–	–	90
<i>Power generation</i>	17.3 (Si)	35	17.3
<i>Power collection</i>	93.7 x 99.95 x 96.3	57	93
<i>Power storage</i>	–		–
<i>DC–microwave conversion</i>	85	85	75
<i>Antenna efficiency</i>	96.53		
<i>Microwave propagation</i>	98	97	98
<i>Rectenna power collection</i>	88	90	76
<i>Microwave–DC conversion</i>	89	85	
<i>DC–Grid</i>	97		75
<i>Total</i>	9.53	12.6	7.7

► *Considering that the average intensity of solar power in space is 5 to 10 times larger than that on the ground, the total efficiency of space solar power generation is higher than that of ground power generation.*

# Is Solar Power Satellite System clean?

*–CO<sub>2</sub> emission compared with existing power plants–*

*(g-CO<sub>2</sub>/kWh)*

<i>Power Plant</i>	<i>Operation Phase</i>	<i>Construction Phase</i>	<i>Total</i>
<i>Space Solar Power Systems</i>	<i>0</i>	<i>20</i>	<i>20</i>
<i>Coal Fired Power Plant</i>	<i>1222</i>	<i>3</i>	<i>1225</i>
<i>Oil Fired Power Plant</i>	<i>844</i>	<i>2</i>	<i>846</i>
<i>LNG Fired Power Plant</i>	<i>629</i>	<i>2</i>	<i>631</i>
<i>Nuclear Power Plant</i>	<i>19</i>	<i>3</i>	<i>22</i>

*Yoshioka et al., 1999*

- ▶ *It has been analyzed that CO<sub>2</sub> load of Solar Power Satellite System is only a few percent of the power plant using fossil fuels.*

## *Does SPS Produce Sufficient Energy?*

### *– Calculation of energy payback time –*

<i>System</i>	<i>SPS</i>		<i>Power Station on ground</i>
<i>Location of Power Module Production</i>	<i>On the Ground</i>	<i>On the Moon</i>	<i>On the Ground</i>
<i>Input Energy [10<sup>9</sup>MJ] (A)</i>	<i>53</i>	<i>37</i>	<i>8. 2</i>
<i>Production of Module</i>	<i>22</i>	<i>22</i>	<i>8. 2</i>
<i>Transportation of Module</i>	<i>31</i>	<i>5.1</i>	<i>–</i>
<i>Power Generation [10<sup>9</sup>Wh/year]</i>	<i>7.88</i>	<i>7.88</i>	<i>1. 23</i>
<i>Energy [10<sup>9</sup>MJ/year] (B)</i>	<i>76.7</i>	<i>76.7</i>	<i>12. 0</i>
<i>EPT [year] (=A/B)</i>	<i>0.69</i>	<i>0.35</i>	<i>0. 68</i>

*“Energy payback time” is an index of how many years it takes to recover the energy required to construct the energy system.*

*► The energy payback time of SPS is analyzed to be less than one year, which is sufficiently small compared to the life of SPS (about 40 years).*

## *Is SPS economically viable?*

### *–Cost estimation for major SPS models–*

<i>Model</i>	<i>Report</i>	<i>Life years</i>	<i>Total Cost</i>	<i>Power Cost</i>	<i>EPT years</i>
<i>NASA Reference System</i>	<i>1980</i>	<i>30</i>	<i>12 B \$ (1977 \$ base)</i>	<i>4 ¢/kwh</i>	<i>–</i>
<i>NEDO Grand Design</i>	<i>1994</i>	<i>30</i>	<i>2,400 B ¥</i>	<i>23 ¥/kwh</i>	<i>2</i>
<i>NASA Fresh Look Study</i> <i>Sun Tower: ST, Solar Disc; SD</i>	<i>1997</i>	<i>40</i>	<i>ST:35~40 B \$ (3.5~4GW)</i> <i>SD:150 B \$ (30GW)</i>	<i>ST:4 ¢/kwh</i> <i>SD:2 ¢/kwh</i>	<i>–</i>
<i>NASDA 1998 Model</i>	<i>1999</i>	<i>30</i>	<i>2,700 B ¥</i>	<i>23.2 ¥/kwh</i>	<i>5</i>
<i>JAXA 2004 Model</i>	<i>2005</i>	<i>40</i>	<i>1,260 B ¥</i>	<i>8.6 ¥/kwh</i>	<i>1.23</i>
<i>USEF Multi-Tether Model</i>	<i>2008</i>	<i>40</i>	<i>1,300~2,100 B ¥</i>	<i>9.3~15.9 ¥/kwh</i>	<i>–</i>
<i>SPS-ALPHA(DRM-4)</i>	<i>2012</i>	<i>&gt;30</i>	<i>12.2 B \$</i>	<i>15 ¢/kwh</i>	<i>–</i>

► *It is analyzed that the power cost of SPS can be close to the power cost of the other ground power plants by using commercial products and reducing the space transportation cost to about 1/50 of the current cost.*

# Comparison between solar power plant on ground and Solar Power Satellite

<i>Subject</i>	<i>Solar Plant on Ground</i>	<i>Solar Power Satellite (SPS)</i>
<i>Solar Power Density (Time Average)</i>	<i>140 W/m<sup>2</sup> (Tokyo)</i>	<i>1370 W/m<sup>2</sup> (Sun-pointing Type) 880 W/m<sup>2</sup> (Earth-pointing, Power Generation on Both Sides)</i>
<i>System Configuration</i>	<i>Solar Cells-Power Collection/Conversion</i>	<i>Solar Cells-Power Collection/Conversion in Space-Wireless Power Transmission-Power Collection/Conversion on Ground</i>
<i>Ratio of Power Generation Efficiency(per Unit Area) (Average Solar Power Density) x (System Efficiency)</i>	<i>1</i>	<i>6~2 (60 % : Efficiency of Wireless Power Transmission)</i>
<i>System Life</i>	<i>30 Years (Weatherability)</i>	<i>30~40 Years (Space Radiation, Debris Impact)</i>
<i>Operation</i>	<i>Power System</i>	<i>Power System &amp; Spacecraft</i>
<i>Ratio of Land Use Efficiency</i>	<i>1</i>	<i>2 (Tether SPS) , 7 (JAXA M-SPS)</i>
	<i>Sun light is Shielded by Solar Panel.</i>	<i>Sun Light Transmission (60-80%)</i>
<i>Power Supply</i>	<i>Fixed, Dependent on Power Grid</i>	<i>Switchable, Using Wireless Power Transmission</i>
<i>Power Fluctuations</i>	<i>Weather, Day &amp; Night</i>	<i>Constant (Sun Pointing Type), Local Time Variation (Earth Pointing Type)</i>
<i>Constraints in Scale</i>	<i>Limitation in Land Use</i>	<i>Essentially no limitation</i>
<i>Positioning in Future Society</i>	<i>Extension of Existing Technology</i>	<i>Paradigm Shift (Advance into Space)</i>
<i>Energy Payback Time</i>	<i>1-3 Years</i>	<i>0.69~5 Years</i>
<i>CO<sub>2</sub> Load</i>	<i>53 g/kWh</i>	<i>20 g/kWh</i>
<i>Electricity Cost</i>	<i>14 Yen/kWh (2017) 7 Yen/kWh (2025) (Goal by NEDO)</i>	<i>4~23 Yen/kWh</i>

► SPS is superior to terrestrial solar power plant in many respects.

Fill cell shows superiority. 16



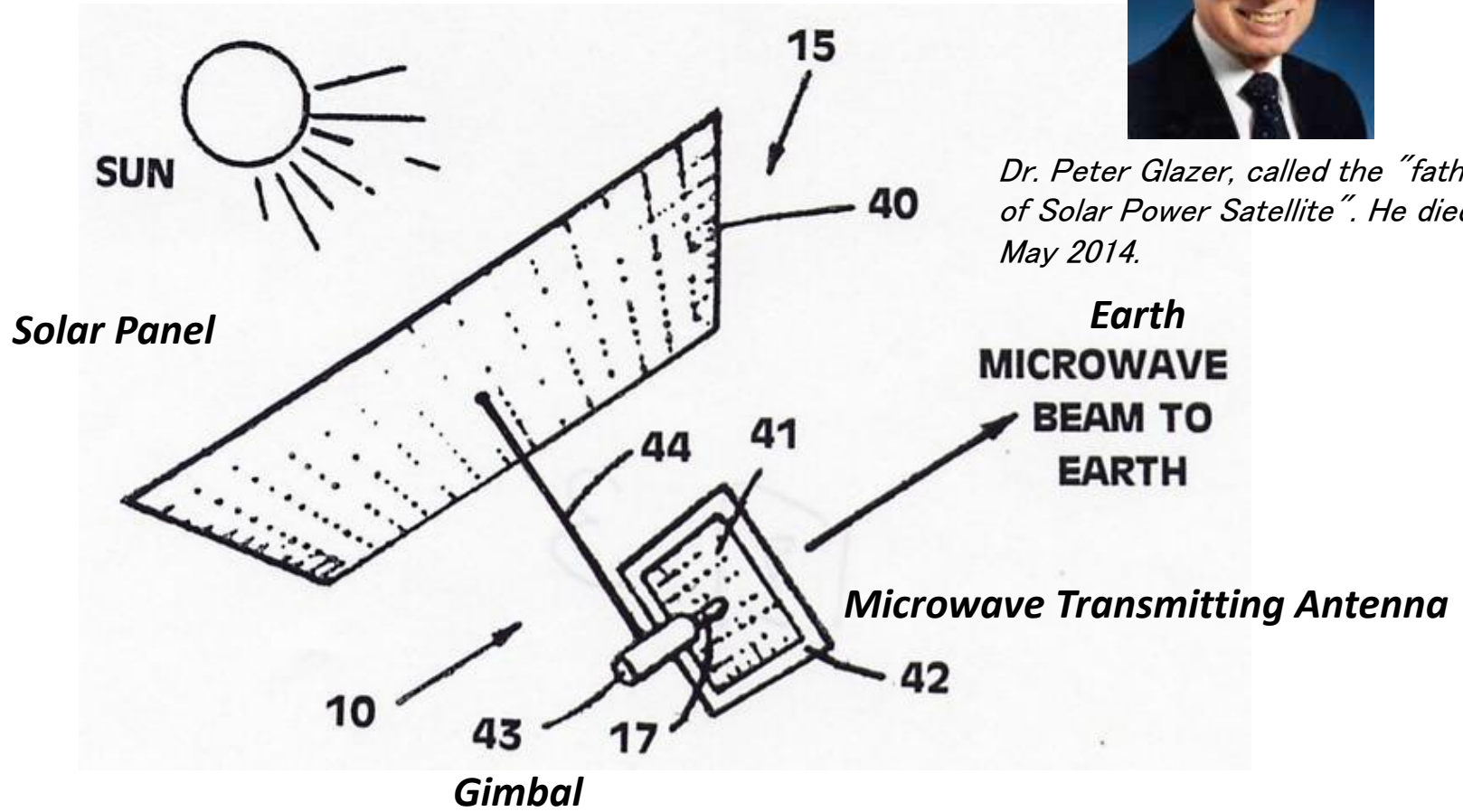
### ***3. Research history***



# *Inventor of Solar Power Satellite*



*Dr. Peter Glazer, called the "father of Solar Power Satellite". He died in May 2014.*



*Patent Announcement by Peter Glazer (1973)*

# Typical SPS Models

## Solar Power Satellite

Microwave

Laser

Non-concentrator

Concentrator

Bus Power

Distributed Power

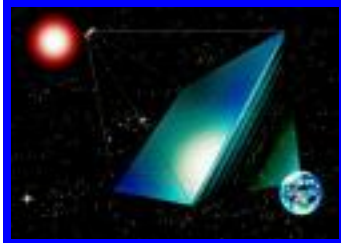
Bus Power

Distributed Power

Laser Direct Excitation



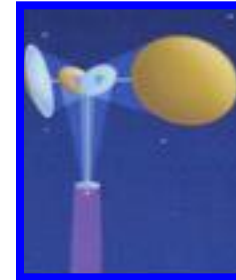
NASA Reference Model



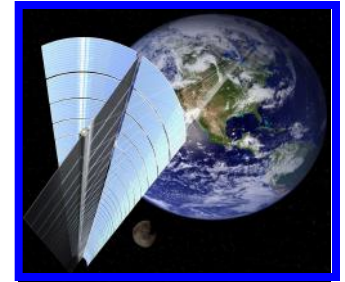
USEF Tether SSPS



NASA Sun Tower



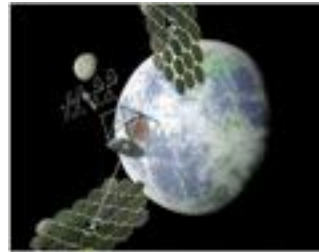
NASDA 2001



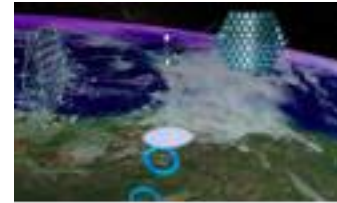
JAXA L-SSPS



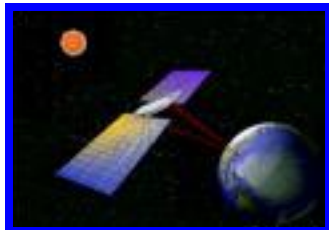
SPS2000



NASA ISC



IAA Study Model



NEDO Grand Design



JAXA M-SSPS



SPS-ALPHA

Designed in Japan

# Which is better, microwave or laser, for SPS?

<b>Transmission Band</b>	<b>Microwave</b>	<b>Laser</b>
<b>Frequency/Wave Length</b>	<b>~several GHz</b>	<b>~1 <math>\mu\text{m}</math></b>
<b>Power Conversion</b>	<b>Solar-DC-RF . . . DC</b>	<b>Solar-DC-Laser- . . . DC Solar-Laser . . . DC</b>
<b>Conversion Efficiency</b>	<b>Higher</b>	<b>Lower</b>
<b>System Size</b>	<b>Larger</b>	<b>Smaller</b>
<b>Beam Energy Density</b>	<b>Lower(Safer)</b>	<b>Higher(smaller beam size)</b>
<b>Electromagnetic Compatibility</b>	<b>Lower</b>	<b>Higher</b>
<b>Weather Dependence</b>	<b>Smaller (typically 97% transmission)</b>	<b>Larger (typically 35-40 % transmission)</b>
<b>Technology Maturity</b>	<b>Higher</b>	<b>Lower</b>
<b>Application</b>	<b>Space-Ground</b>	<b>Space-Space Space-Ground</b>

*Fill cell shows superiority.*

► *Microwave is considered to be more advantageous than laser for wireless power transmission from space to ground.*

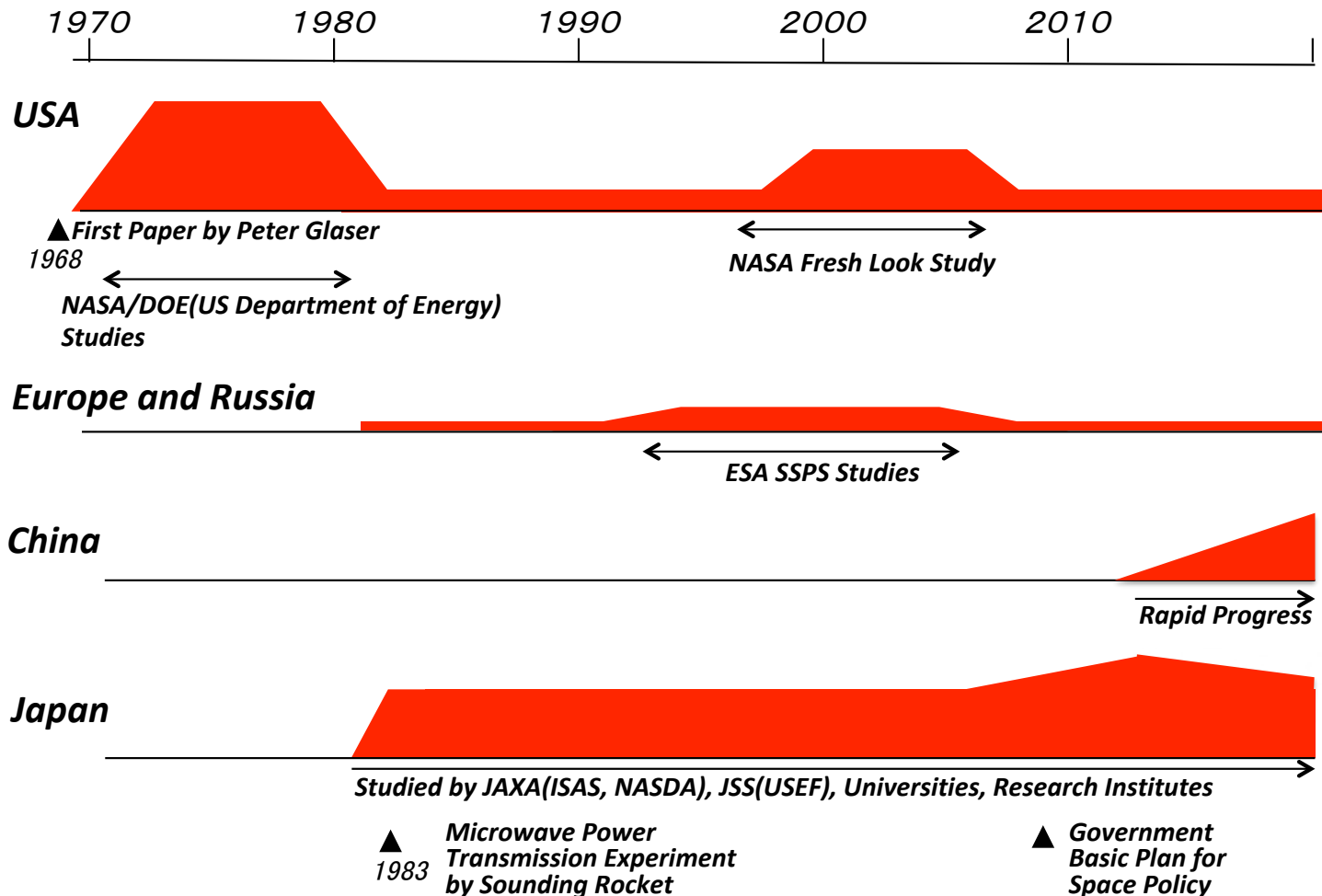
## Output Power, Total Weight, and Specific Weight of Typical SPS Models

<i>Phase</i>	<i>Model</i>	<i>Output Power</i>	<i>Weight</i>	<i>Specific Weight</i>
<i>Commercial</i>	<i>NASA Reference Model</i>	<i>6.5 GW</i>	<i>50,000 Tons</i>	<i>7.7 g/W</i>
<i>Commercial</i>	<i>NEDO Grand Design</i>	<i>1.3 GW</i>	<i>20,000 Tons</i>	<i>15.3 g/W</i>
<i>Commercial</i>	<i>JAXA M-SSPS Model</i>	<i>1.34 GW</i>	<i>9,907 Tons</i>	<i>7.4 g/W</i>
<i>Commercial</i>	<i>USEF Tethered SPS</i>	<i>1.32 GW</i>	<i>26,500 Tons</i>	<i>20 g/W</i>
<i>Commercial</i>	<i>SPS-ALPHA</i>	<i>500 MW (Ground Output Power)</i>	<i>11,800 Tons</i>	<i>23.6 g/W (per Ground Output Power)</i>
<i>Commercial</i>	<i>Sun Tower(GEO)</i>	<i>1.2 GW</i>	<i>22,300 Tons</i>	<i>19 g/W</i>
<i>Commercial</i>	<i>Integrated Symmetrical Concentrator</i>	<i>1.2 GW</i>	<i>18,000~31,500Tons</i>	<i>15g/W~26g/W</i>
<i>Commercial</i>	<i>European Sail Tower</i>	<i>275 MW</i>	<i>2,140 Tons</i>	<i>7.8 g/W</i>
<i>Demonstration</i>	<i>SPS2000</i>	<i>10 MW</i>	<i>240 Tons</i>	<i>24 g/W</i>
<i>Experiment</i>	<i>Tethered SPS Experiment Model</i>	<i>420kW</i>	<i>18.1 Tons</i>	<i>43g/W</i>
<i>Experiment</i>	<i>Small Scale Experiment Model</i>	<i>3.8kW</i>	<i>500 kg</i>	<i>132g/W</i>

► *Specific weight of commercial model is approximately 10 g/W to 20 g/W. The experimental model in the near future has a larger specific weight.*

# International Research Status

The international research activities on Solar Power Satellite is shown below. The height of red area indicates the research activities qualitatively. The research started almost 50 years ago in the USA. Now, Japan is the leading country in the research and development. China is rushing lately.



# ***SPS Major History (International)***

- 1968 *First concept by Peter Glaser "Power from the Sun: its Future" in "Science (Journal)"*
- 1970' *NASA/DOE (Department of Energy) joint study*
- 1980 *Study in US terminated by President Reagan*
- 1983 *First sounding rocket experiment in Japan to study microwave/plasma Interaction*
- 1990 *ISAS study "Demonstration Model SPS 2000" in Japan*
- 1995 *NASA study (Flesh Look, SCDS, SERT, SCTM) (-2004)*
- 1998 - *NASDA (now JAXA) SSPS study in Japan*
- 2000 - *USEF (now Japan Space Systems) SSPS study in Japan*
- 2002 - 2004 *ESA SPS study*
- 2009 *SPS research included in Government Basic Plan for Space Policy in Japan*
- 2010 - *China SPS study*
- 2011 *First international assessment of SPS by International Academy of Astronautics*

# SPS Major History (United States)

- 1970s NASA/DOE (Department of Energy) joint study  
1977-1980 NASA Conceptual Design (20 M\$)  
1978 DOE's SPS Concept Development and Evaluation Program(CDEP)
- 1980 Because the construction cost and the power price were considered very high, systematic study was suspended due to NAS (National Academy of Sciences) evaluation (freezing of SPS R & D investment) and the Reagan administration's austerity policy.
- 1995 NASA study resumption  
1995-1997 Fresh Look Study (30 SSP system concepts)  
1998 SSP Concept Definition Study (2 M\$)  
1999-2000 SSP Exploratory Research and Technology (SERT) program(2.2 M\$)  
2001 US National Research Council(NRC) Report on SSP Strategic research and technology road map  
2001-2002 SSP Concept and Technology Maturation (SCTM) program (3 M\$)  
2001-  
-2004 Budgeted reduction  
Joint investment in SSPS technology research and studies by NASA and NSF(National Science Foundation)
- 2004 "New Space Exploration Program" by Bush Administration  
SPS-related technology development was incorporated into research and development as part of advanced technology.
- 2005-08 Low level NASA budget for SPS
- 2007 Pentagon (National Security Space Office) investigated Space Solar Power from the perspective of securing power in combat units and disaster areas (M.Smith (NSSO), J.Mankins, D.Preble).
- 2008 The longest microwave power transmission experiment (160km) in history using the retro directive system in Hawaii. Under the planning of the US team, Kobe University was in charge of power transmission technology. It aired on Discovery Cannel. A WPT demonstration on the ISS was considered. NASA JSC lead the study. DOD and universities participated, but the study was terminated in nine months.
- 2010 The Obama administration proposed a budget that emphasizes advanced technology development.
- 2011 SPS was selected as one of the research themes of NIAC (NASA Innovative Advanced Concepts). However, the budget for the following project was not available
- 2015 SPS research agreement signed with California Institute of Technology Caltech) sponsored by Northrop Grumman Corporation
- 2019 Laser power transfer experiment at US Naval Research Laboratory (2kW, 325m)

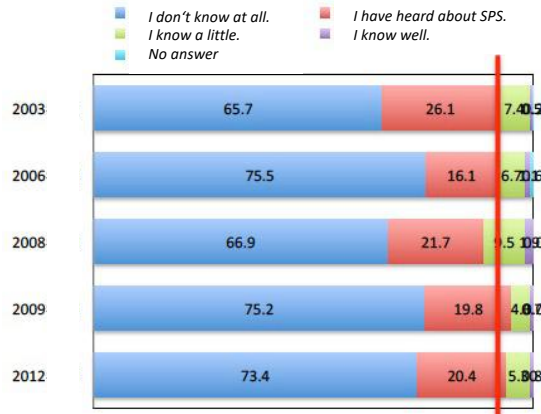


# History of SPS Research in Japan

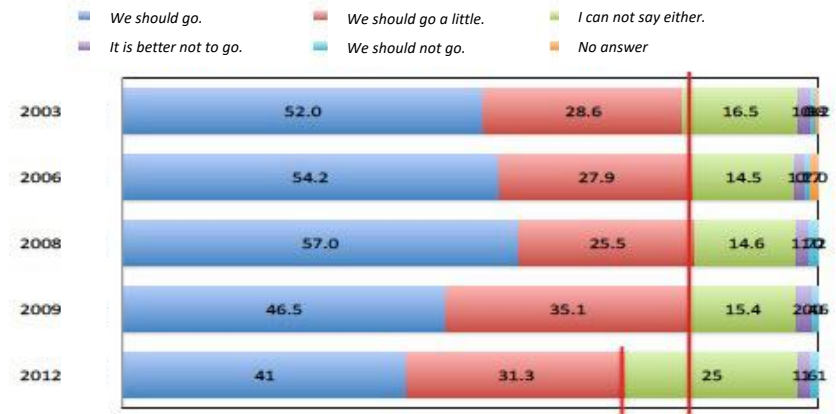
- 1983 *MINIX* experiment (Space experiment to study the interaction of microwave with the ionosphere by sounding rocket) (2.45GHz)
- 1987 SPS Working Group was organized at ISAS and started design of SPS demonstration model SPS2000
- 1992 *MILAX* Airplane experiment (microwave power transmission to small airplane) (2.45GHz)
- 1993 *ISY-METS* experiment (space experiment to study the interaction of microwave with the ionosphere by sounding rocket) (2.45GHz)
- 1994~ Microwave garden experiment (microwave exposure experiment for plant growth, Advanced Industrial Science and Technology ) (2.45, 5.8GHz)
- 1995 *ETHER* experiment ( microwave power transmission to an airship) (2.45GHz)
- 1997 Space Solar Power Research Society (SSPRS) in Japan was organized.
- 1998 ~ NASDA (JAXA) SPS Study
- 2000 ~ USEF/METI (Ministry of Economy, Trade and Industry) SPS Study
- 2006 Sounding rocket experiment for antenna deploy and active phased array
- 2009 Microwave power transmission experiment from an airship(2.45GHz)
- 2009 SPS research was included in Government Basic Plan for Space Policy.
- 2010 All Japan project (USEF/JAXA) for microwave power transmission experiment started.
- 2014 Space Solar Power Systems Society was organized by the evolutionary dissolution of Space Solar Power Research Society.
- 2015 1.8 kW Microwave power transmission experiment (55 m, JSS/JAXA)  
10 kW Microwave power transmission experiment (500 m, JSS)
- 2016 Laser power transmission experiment in vertical direction (JAXA)
- 2019 Microwave power transmission experiment to drone (JSS/JAXA)

# Social Awareness and Reality of SPS

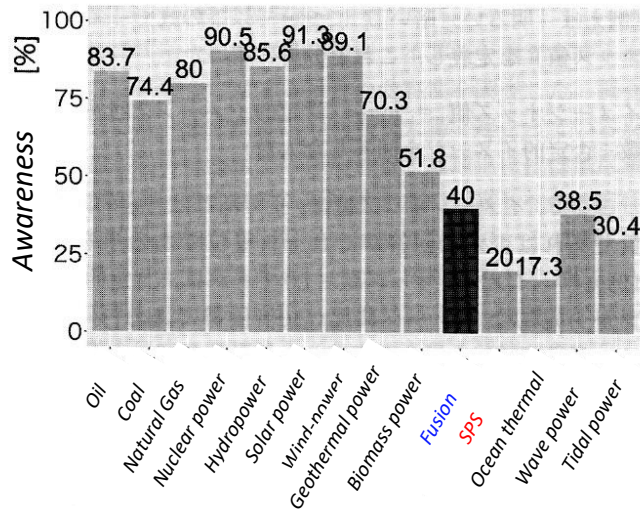
## Do you know SPS?



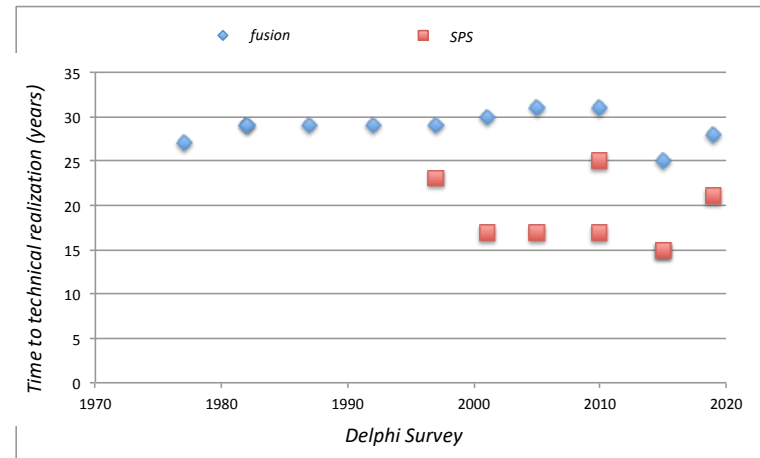
## Do you think we should proceed with this future system?



► Results of five surveys conducted by JAXA from 2003 to 2012 on the awareness and necessity of SPS. Little is known to the general public (65–75% did not know), but they are positive about the concept.



► Awareness of various energy sources in questionnaires conducted by fusion researchers. Compared to nuclear fusion, SPS was less well known.



► Comparison of the time required to realize SPS and fusion (excluding inertial fusion) in the Delphi survey. Scientific research experts believe that SPS will be realized sooner than fusion.

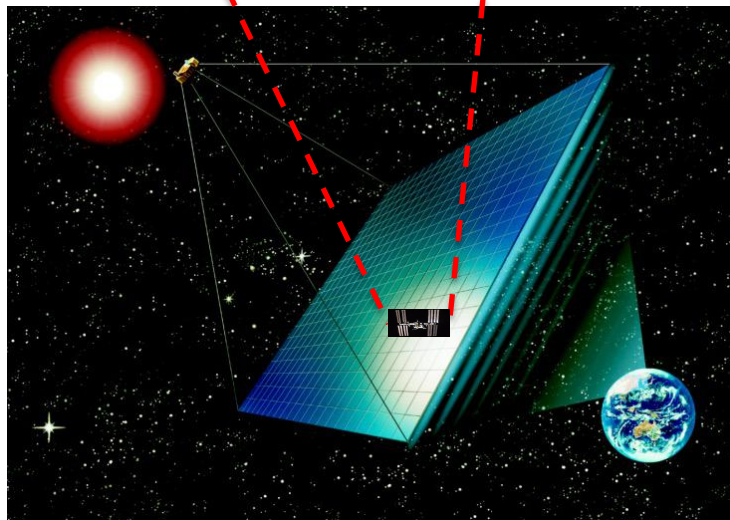
# ***4. Required technologies***



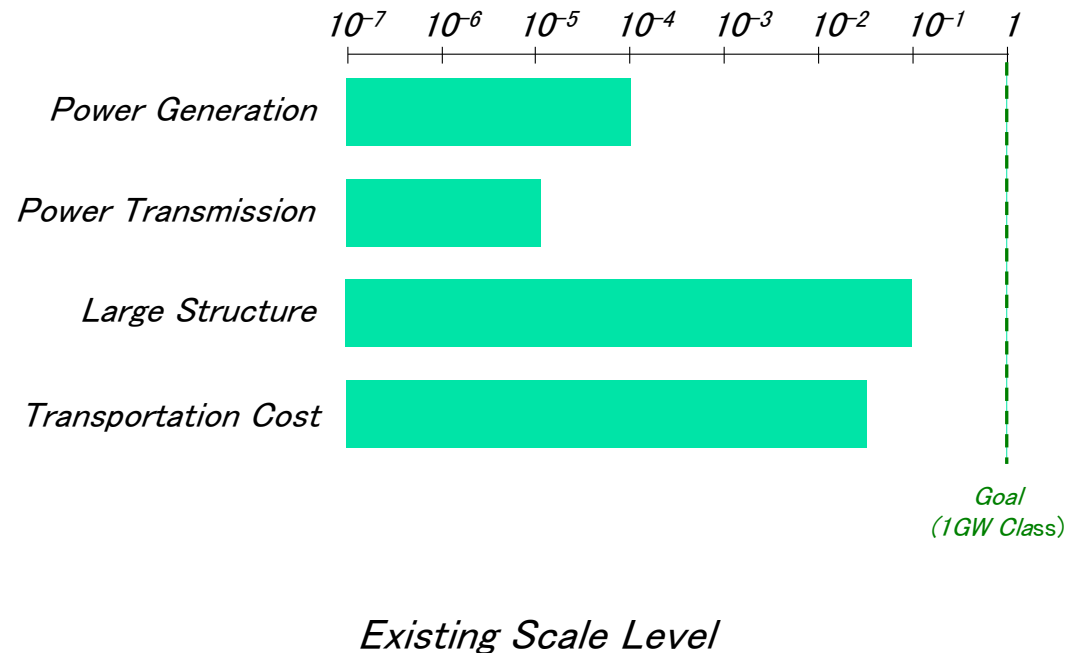
# Current and Target Level for SPS Technologies

<i>Primary technology</i>	<i>Existing level</i>	<i>Target level</i>	<i>Order of magnitude</i>
<i>Solar power generation</i>	<i>100 kW(space)</i>	<i>1 GW</i>	<i>10,000</i>
<i>Microwave power transmission</i>	<i>10 kW (ground)</i>	<i>1 GW</i>	<i>100,000</i>
<i>Large space structure</i>	<i>100 m (space)</i>	<i>1 km</i>	<i>10</i>
<i>Space transportation</i>	<i>5,000 \$/kg</i>	<i>100 \$/kg</i>	<i>1/50</i>

*ISS 100m Scale*



*SPS 1-2 km Scale*

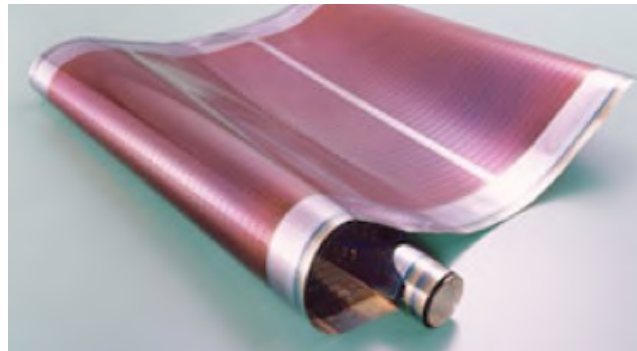


# Power Generation

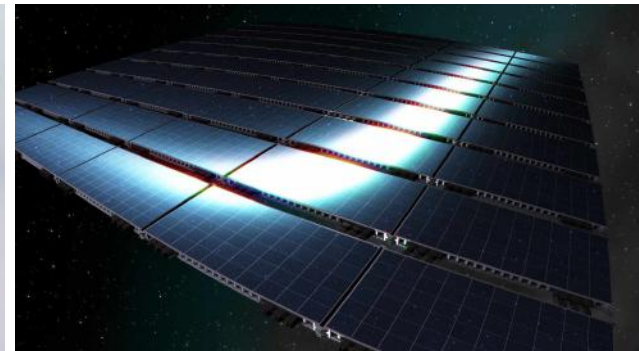
	Current	SPS Target
<b>Conversion Efficiency</b>	15-30 %	35-40 %
<b>Specific Weight</b>	1-100 g/W	1 g/W
<b>Life in Space</b>	10 years	30-40 years
<b>Cost</b>	4-6 \$/W	1-0.5 \$/W



*Conventional solar array panel for space use, with high performance cells, but heavy*



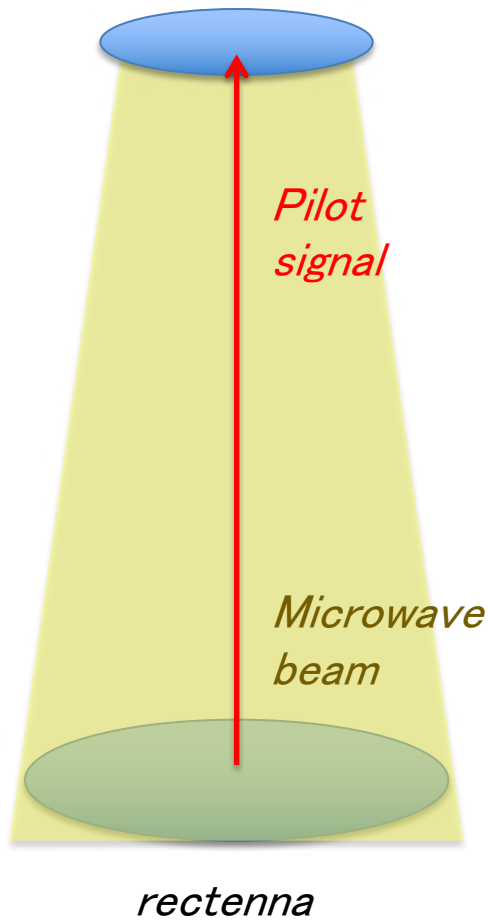
*Light-weight thin film solar cell for ground use, a candidate for SPS*



*Installed on light-weight SPS structure*

# Technologies for Microwave Beam Direction Control

Spacetenna(array antenna)

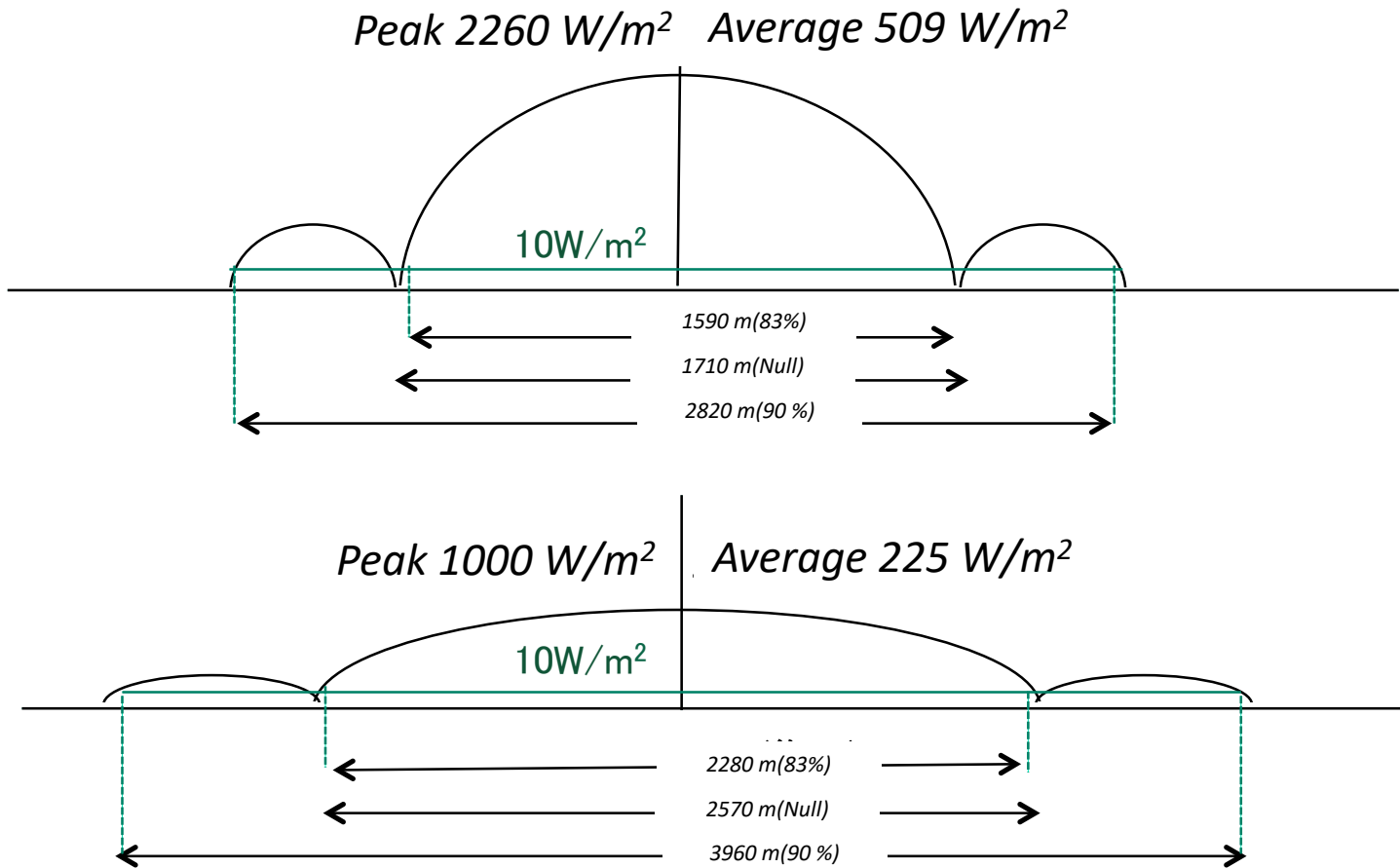


## Required Technologies

1. Determination of rectenna position by detecting and processing pilot signal from the rectenna site.
2. Controlling the phase of microwave from each antenna (or sub-array) so that the microwave beam as a whole directs to the rectenna center precisely.

Item	Target
Beam forming	100 $\mu$ rad
Beam control accuracy	10 $\mu$ rad
Beam control range	$\pm$ 10 degrees(tentative)
Power concentration	90 % inside the rectenna
Response (phase control)	sec scale for small antenna area(m)
	hour scale for large antenna area(100m)

# Microwave Intensity on the Ground (depending on the SPS Model)

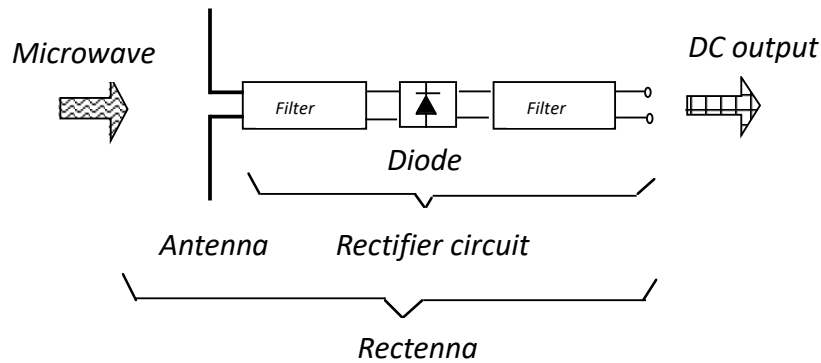


Ground power density for tether SPS with a 2.5 km x 2.375 km antenna (approximated as a circle). 1.4GW transmission power. 37,333km distance (from geostationary satellite orbit to Tokyo).

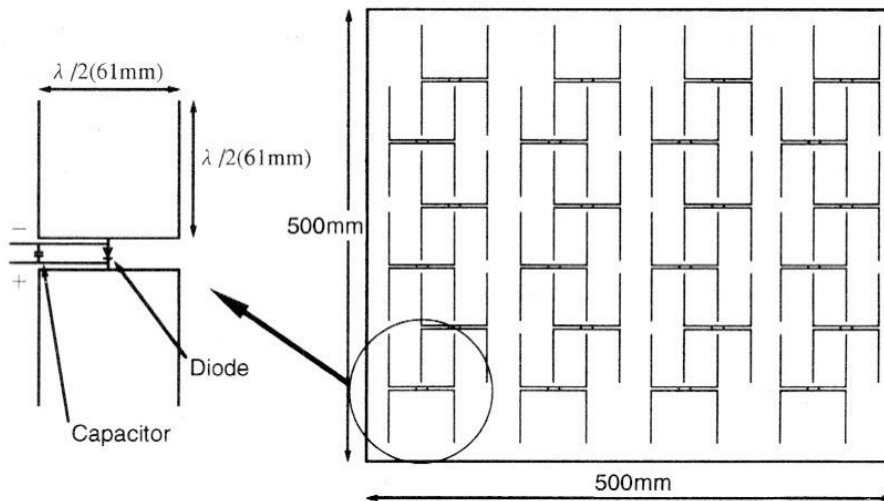
Upper panel: All antenna elements are in the same phase.

Lower panel: The phases of antenna elements are controlled so that the peak intensity is lowered at 1,000 W/m<sup>2</sup>.

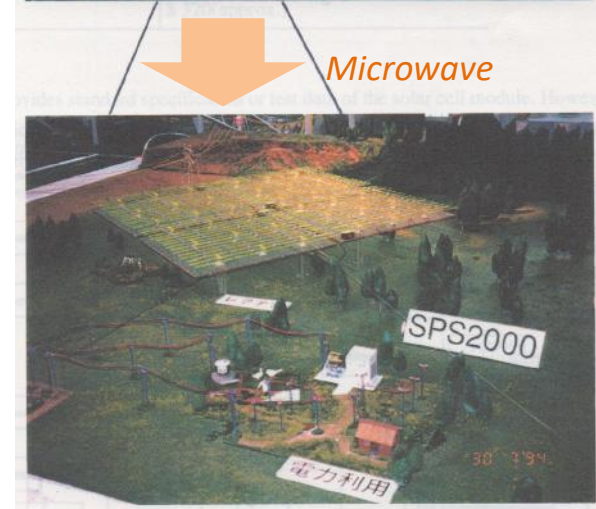
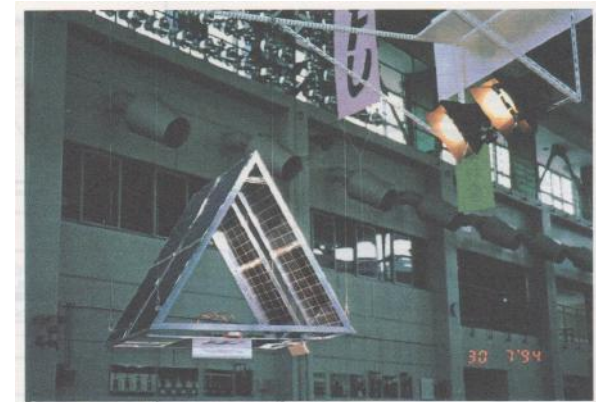
# Rectenna (Microwave Receiving Circuit) Technologies



Rectenna operating principle (Circuit diagram)



Rectenna prototyped in the laboratory

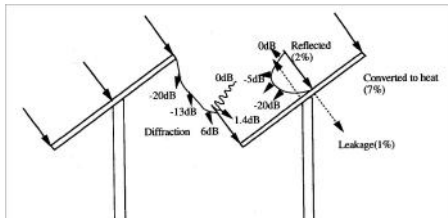


SPS model and rectenna in the laboratory test. Microwaves (2.45GHz) from a model simulating the solar power satellite SPS2000 are received by a rectenna simulating a ground power receiving station. A motor and illuminant are operated by the received power on the ground.

► Many prototypes of small power receiving antennas have already been made and tested. A rectification efficiency of about 80% has been already achieved.



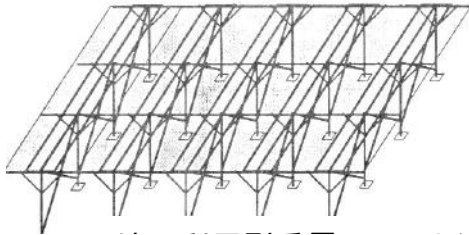
# Microwave Receiving Antenna (Design Examples)



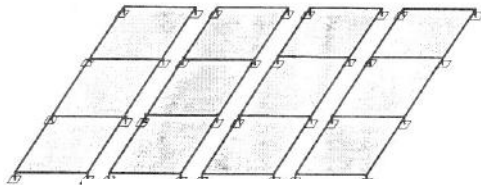
Receiving antenna of NASA Reference System



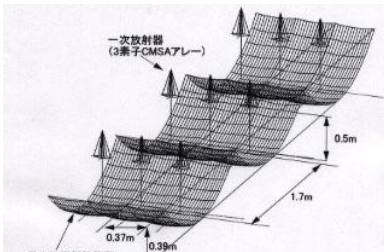
Image of receiving antenna site (NASA Reference System)



Receiving antenna above the ground (SPS2000)



Receiving antenna on the ground (SPS2000)



Parabola type receiving antenna (Tether SPS)

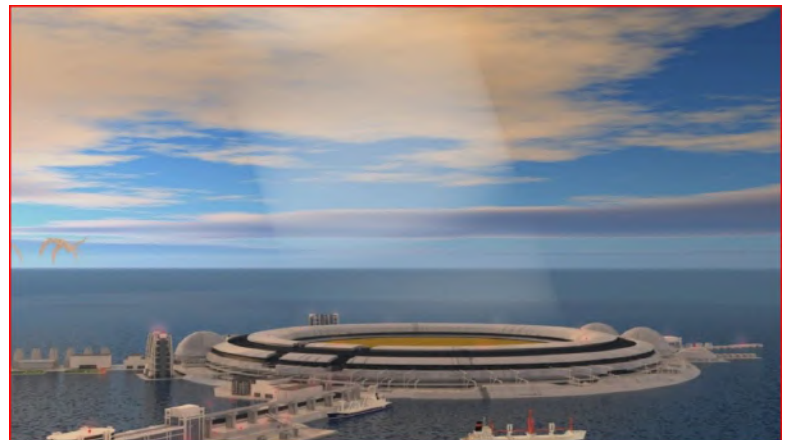
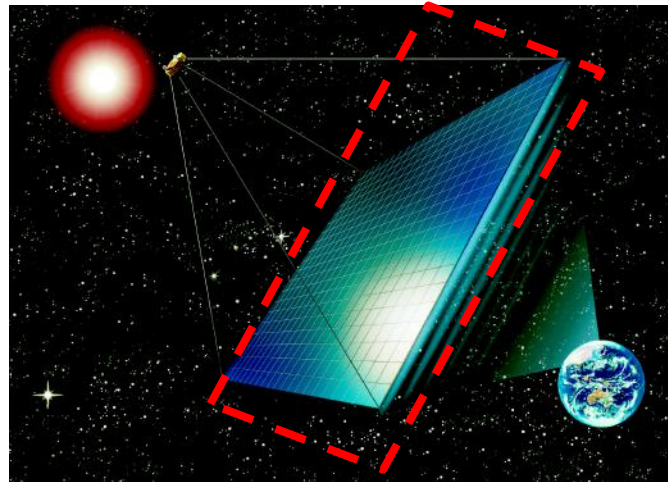


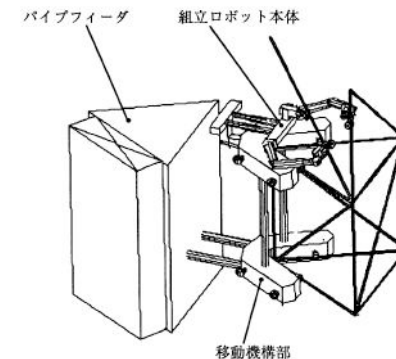
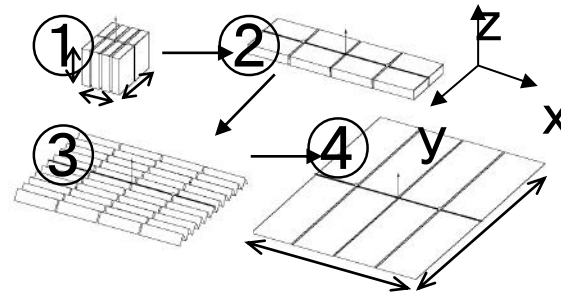
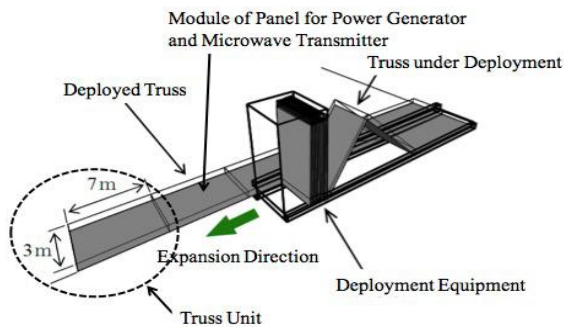
Image of marine receiving antenna site (JAXA)

# Technology for building large structures in space

*1 GW class SPS requires a structure of km size.*



*The basic structure is a large thick panel. Several construction methods have been proposed and prototyped.*



► *Deployment and assembly at the same time*

► *Automatic self-deployment*

► *Construction by robot*<sub>34</sub>

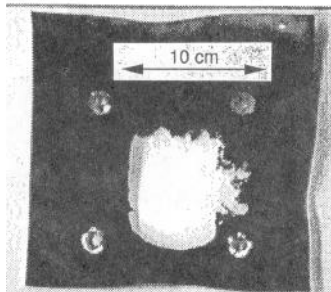
# Will Solar Power Satellite be destroyed by the debris or meteorites impacts?

*Debris or meteorites impacts are inevitable. The structure can be designed to localize the damage so as that the damage is within an allowable range of SPS.*

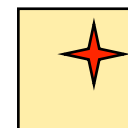
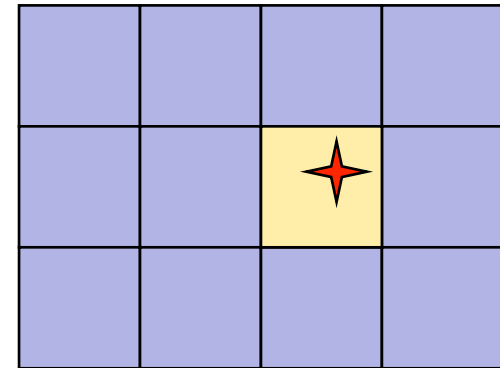
*Collision frequency (impacts/km<sup>2</sup>) for the geosynchronous orbit is:*

*once per 70 years for 10 cm size object,  
once per 3 years for 1 cm size object, and  
2400 times per year for 1mm size object.*

*The solar panel will be designed so as that the impact damage is localized in a module. If we adopt a module size of 0.5 m x 0.5 m, the power loss during 40 years will be 4.8%.*



*An example of damage observed in hyper-velocity impact experiment. The target is a plate simulating the solar panel. The damage size often reaches 10 times larger than the size of the projectile.*



*Failed module  
by impact*



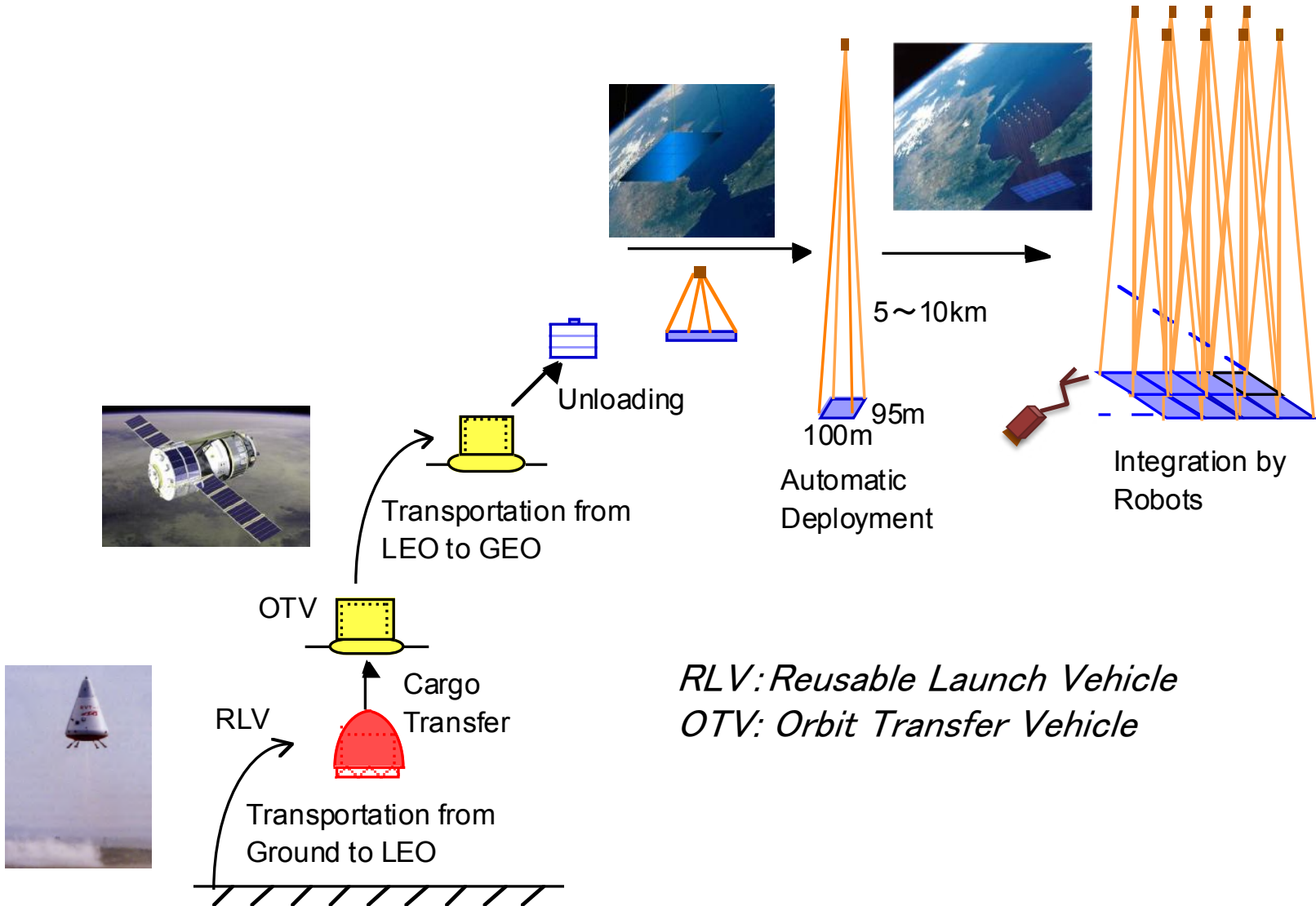
*Good module*

*The solar array panel is designed that the impact damage does not propagate to other modules.*

***5. Required space  
transportation***

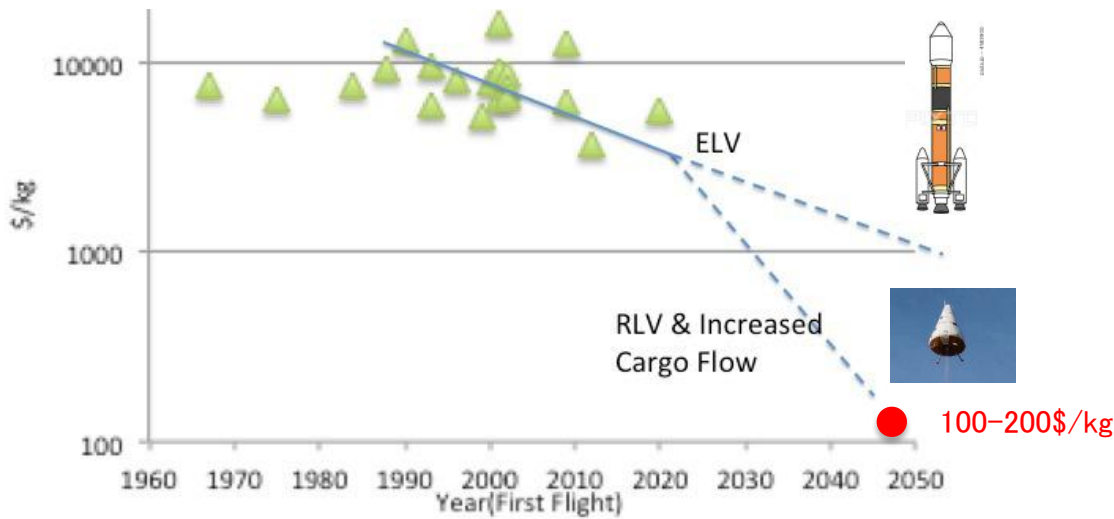


# Typical Construction Scenario (An Example for Tether SPS)

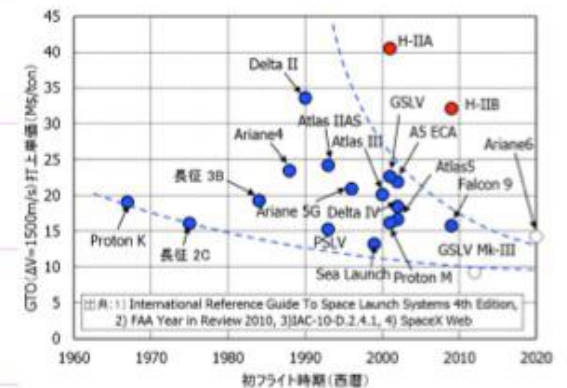
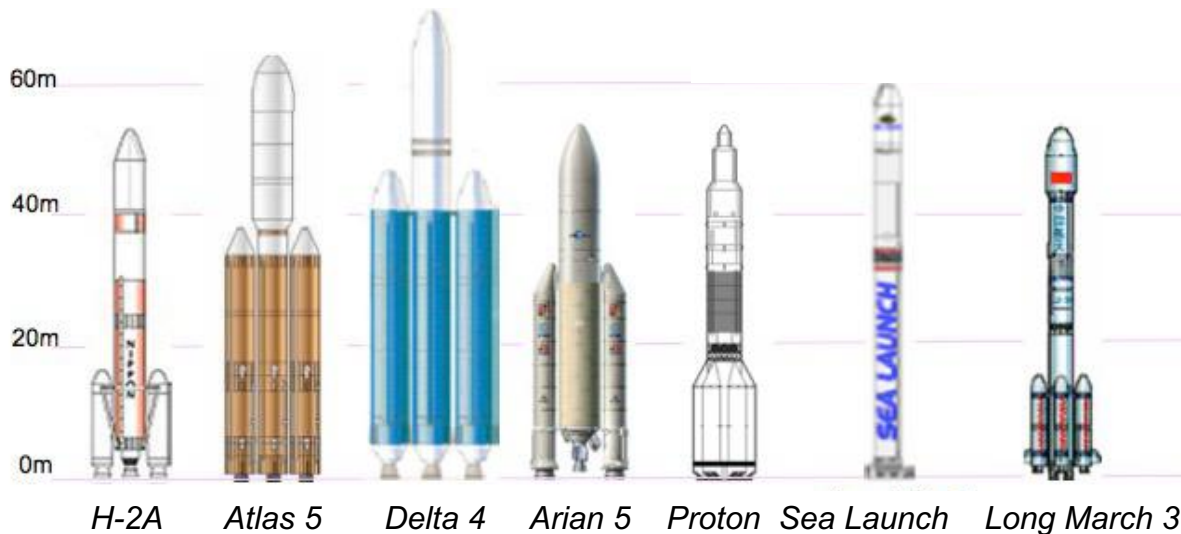


# Transportation Cost for Launch Vehicle

## LEO Launch Cost

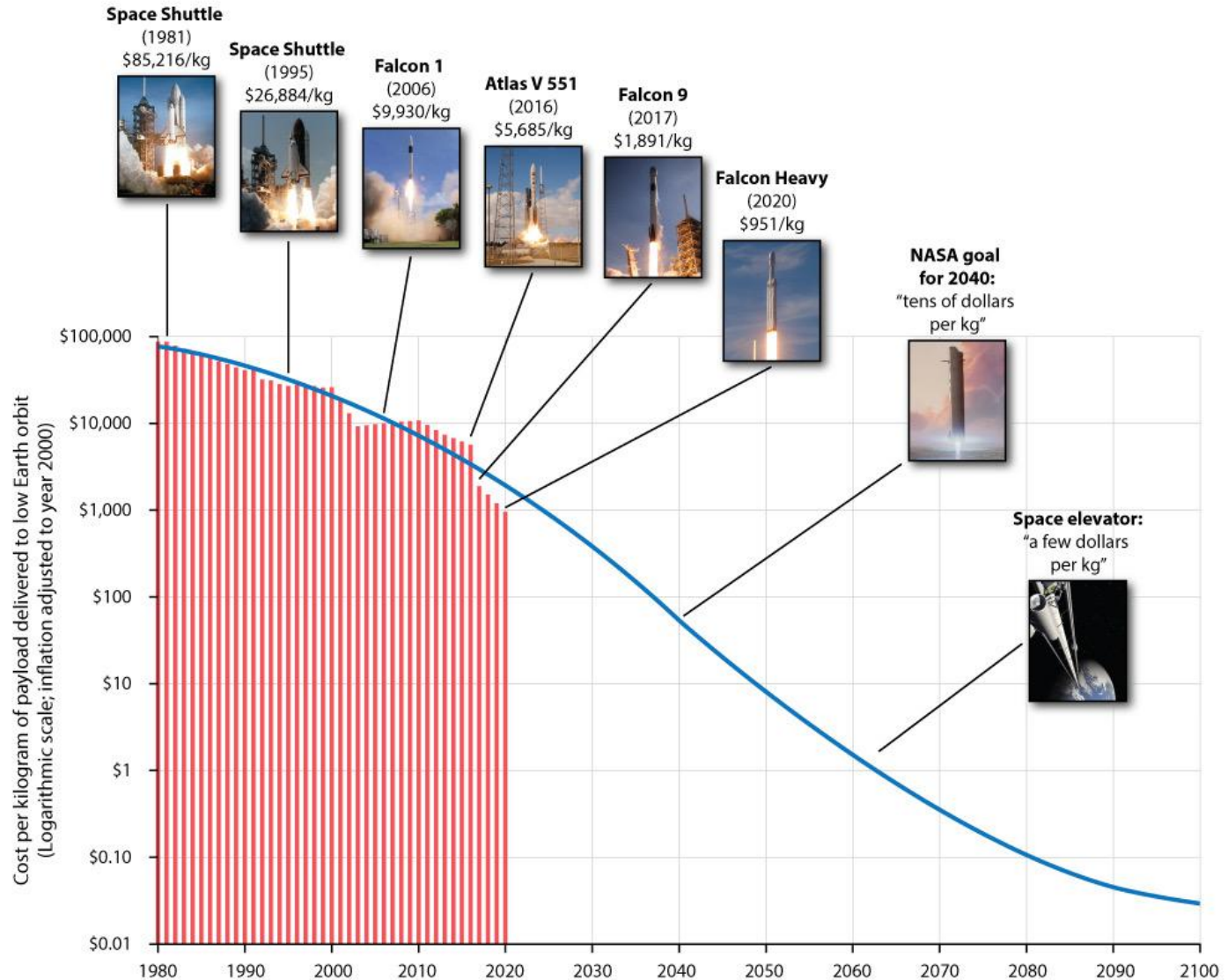


RVT-9 Weight: 500kg, Length: 3.5m



GTO Launch Cost

# Long-term Prediction of space transportation costs



*Prediction by Future Timeline Net*

FutureTimeline.net

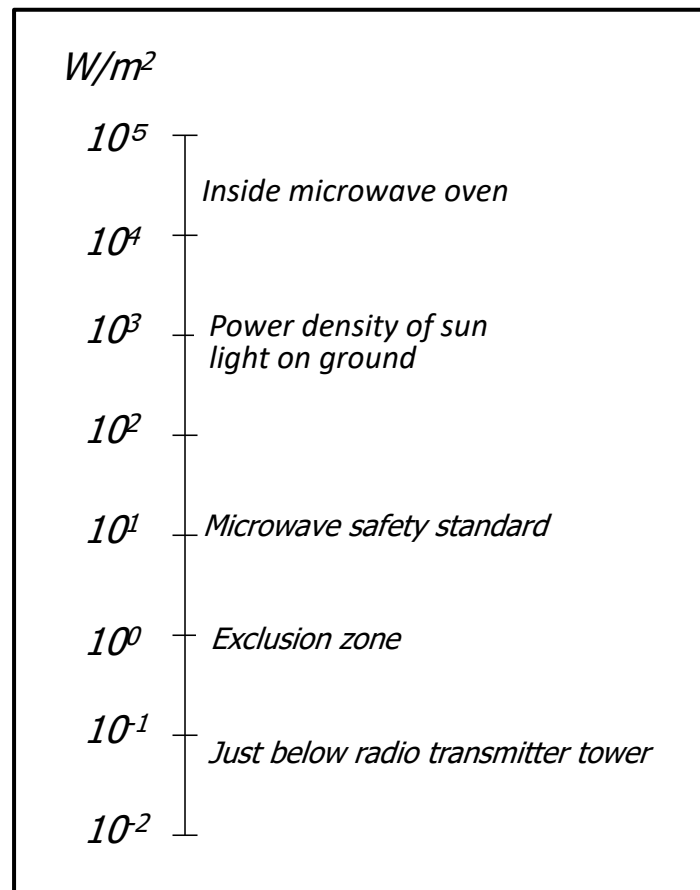


***6. Safety and environmental issues, and legal aspects***



# Microwave Safety Issues

- ▶ In accordance with Japan's and international safety standards, it is considered that areas with strength above the standards are designated as an exclusion zone, and areas up to 1/10 of the standards are restricted areas in principle for safety reasons. The signal strength outside the restricted area is about the same as or even lower than that of mobile phones.
- ▶ Even within the exclusion zone, the maximum power intensity is about the intensity of the sun light, so it is considered that there will be no effect on birds or other animals that temporarily invade.
- ▶ It is possible to control to stop the transmission of microwave power immediately when the microwave is about to deviate from the power receiving station due to some kind of failure.



Agency, Country	Exposure limit in general public		Exposure limit to professional personnel	
	Electric Field Strength (v/m)	Power Density (mW/cm <sup>2</sup> )	Electric Field Strength (v/m)	Power Density (mW/cm <sup>2</sup> )
The Ministry of Internal Affairs and Communications, Japan (1990,1997)	61.4	1 General	137	5 Controlled
ANSI/IEEE (USA) C95.1-1999	—	3.87 Non-controlled	—	10 Controlled
ICNIPP 1998 (International)	61	1 Public	137	5 Controlled

Microwave protection guideline value (5.8 GHz example). The guideline values for Japan and ICNIRP are the same for 2.45 GHz.

# **General ideas on the impact of microwaves on the human body, social infrastructure, and ecology**

*In order to use microwave for wireless power transmission, it is necessary to have the ITU (International Telecommunication Union) assign a new frequency to be used for power transmission business. The basic idea of the effects of microwaves on the human bodies, social infrastructure, and ecology can be summarized as follows.*

## *(1) Effects on human bodies*

*In accordance with international protection guidelines, an area of  $1 \text{ mW/cm}^2$  or more will be an exclusion zone.*

## *(2) Effects on communication infrastructure*

*Since it is difficult to use the same frequency as that already used in the communication infrastructure, it is necessary to receive a dedicated frequency allocation for the wireless power transmission from the ITU. For harmonics, a power transmission system that does not have the same phase in the harmonics shall be used to avoid interference.*

## *(3) Effects on existing infrastructure*

*For aircraft and satellites, filters are used in the onboard electronic devices to prevent interferences during the flight in the microwave beam. No measures will be required for standard electronic devices used on the ground. However, if some special consideration is required for such as medical facilities, the receiving facility will be placed so as to keep a distance from them.*

## *(4) Effects on ecology*

*Regarding the impact on the ecology in the beam (long-term exposure for plants and surface animals with low mobility, short-term exposure for animals with high mobility such as birds and insects), it is considered there will be no problem at the sunlight level ( $100 \text{ mW/cm}^2$ ). Regarding the effect on the ecology under the rectenna, the microwave effect is small considering the power receiving efficiency (80% or more). Rather, it seems that the sunlight blocking effect by the rectenna is important, but it is believed that the ecosystem can be maintained if the rectenna's solar transmittance is 60% or more.*

# **Considerations as an activity principle associated with SPS**

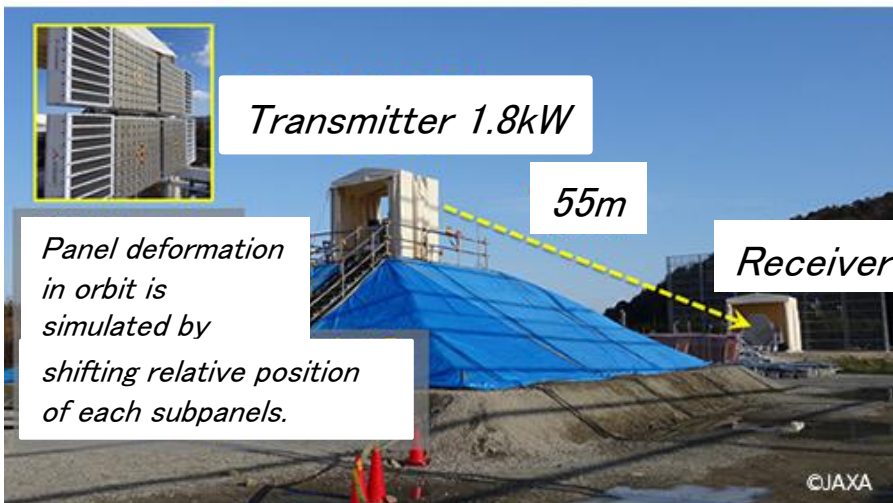
## **(by Prof. Toshio Kosuge)**

- 1. The use of space solar power is the freedom of use of space, and is permitted as long as it complies with space law and general international laws. However, it is necessary to consider the activities and developing countries for the benefit of all countries.*
- 2. Outer space, including the moon and other celestial bodies, is not subject to national acquisition by use, utilization, elections or any other means (Outer Space Treaty Article 2, Moon Treaty Article 11).*
- 3. The treaty, whether by government agencies or non-governmental organizations, have international responsibility for space activities (Article 6 of the Outer Space Treaty).*
- 4. For damages caused by space activities, the parties concerned shall be liable for damages, and in the case of damages on the ground or in the air, they shall be liable for strict liability, and ultimately the nation shall be liable (Space Liability Convention 7). Article, Space Liability Convention).*
- 5. It is not allowed to cause harmful interference to the environmental protection of space, the earth, and the space activities of other parties (Article 9, Moon Treaty, ITU Charter, etc.).*

***7. Current R & D status and  
roadmap for realization***



# Recent microwave power transmission experiments on ground



*Microwave power transmission experiment conducted in 2015 (distance 54m, 5.8GHz, microwave output 1.8kW, power receiving output 340W (JSS / JAXA))*

*Transmitter*



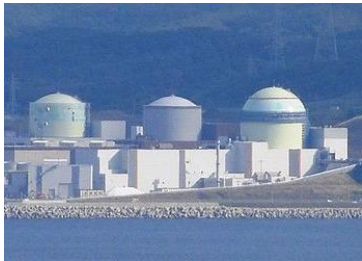
*Receiver*



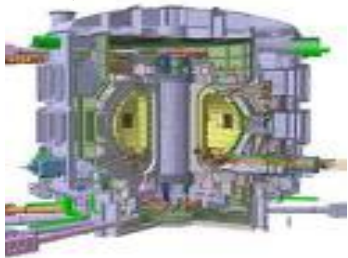
*Microwave power transmission experiment conducted in 2015 (distance 500m, 2.45GHz, microwave output 10kW, power receiving output 32W (JSS))*

# SPS Development Roadmap as Compared with Nuclear Fission and Fusion

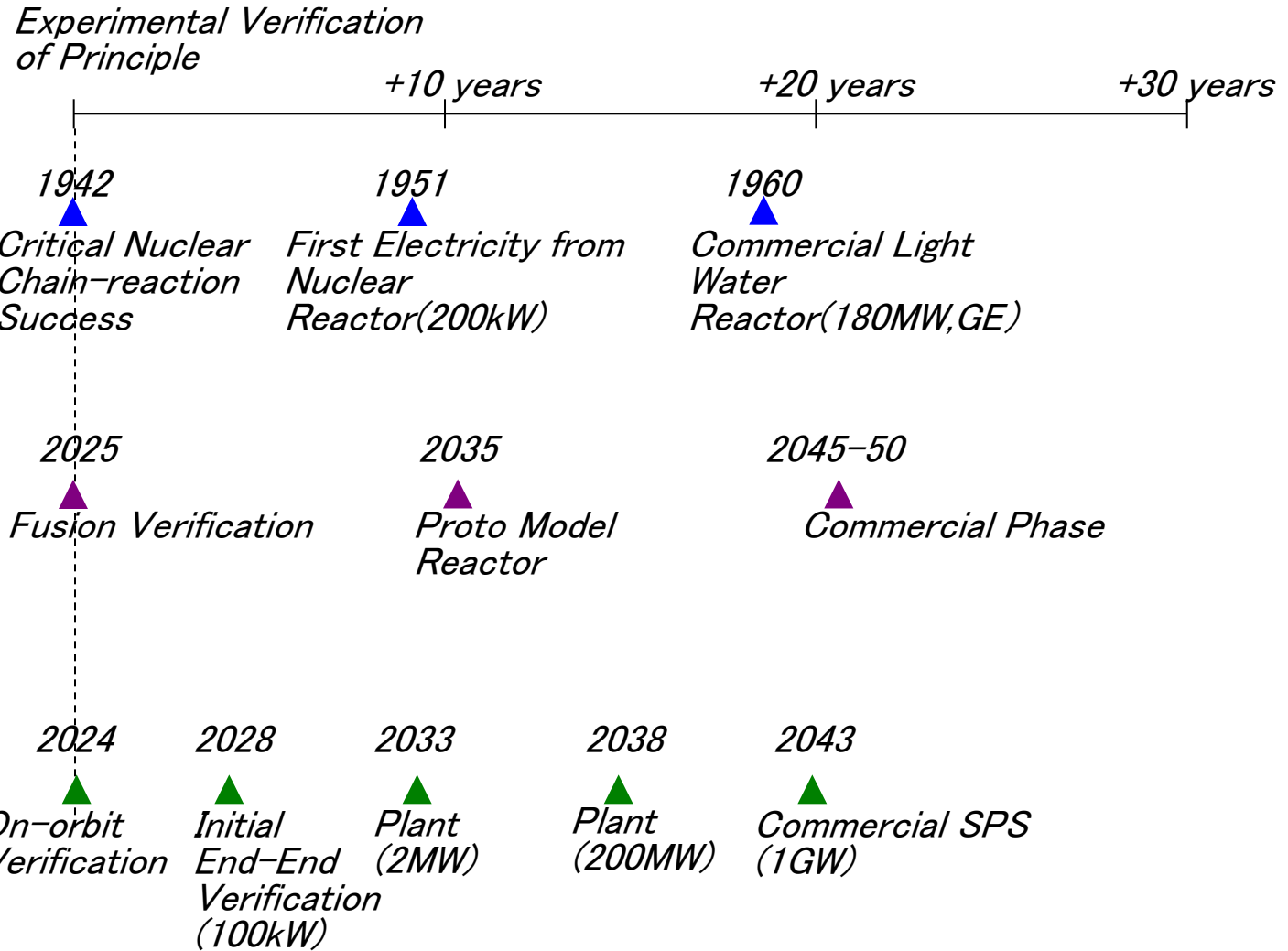
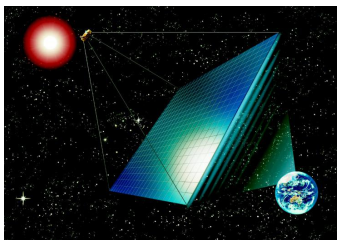
Nuclear Fission Plant (History)



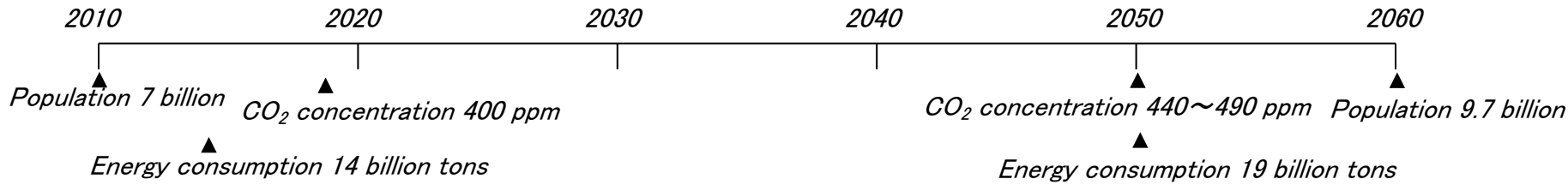
Nuclear Fusion Plant (Plan)



Solar Power Satellite (Plan)



# SPS Research and Development



Research phase

Development phase

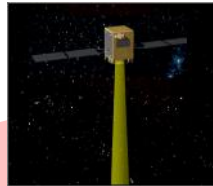
Commercial phase (1GW/year)

Microwave-type

Test on ground



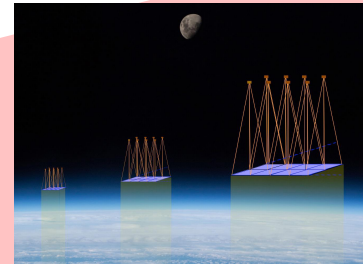
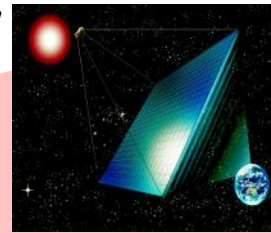
Test in space



Plant for verification

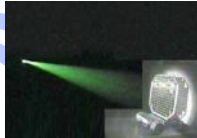


Commercial SPS

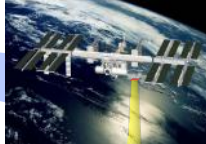


Laser-type

Test on ground



Test in space



Plant for verification



Space-space or space to ground  
Commercial SPS



Rover for planetary exploration



Power station on the Moon



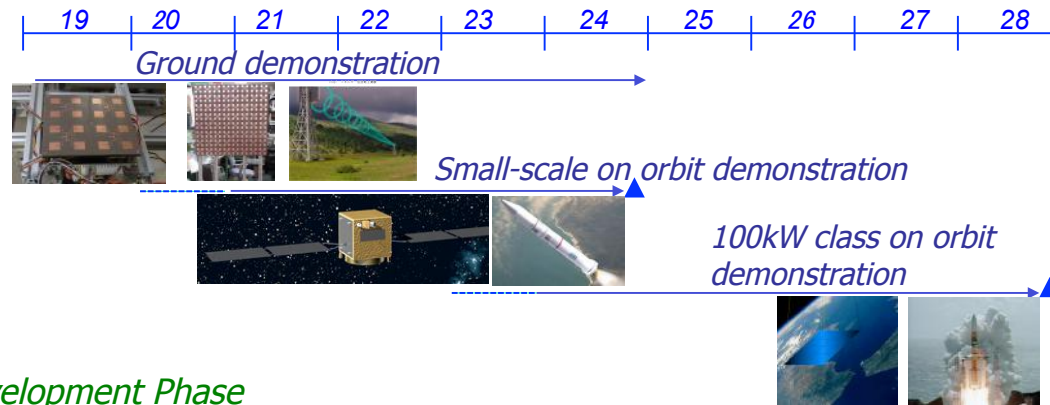
Application in planetary exploration

► Since the laser transmission is strongly affected by the weather, it is more likely to be used for spacecraft-to-spacecraft or lunar-planetary exploration than power transmission to the ground.

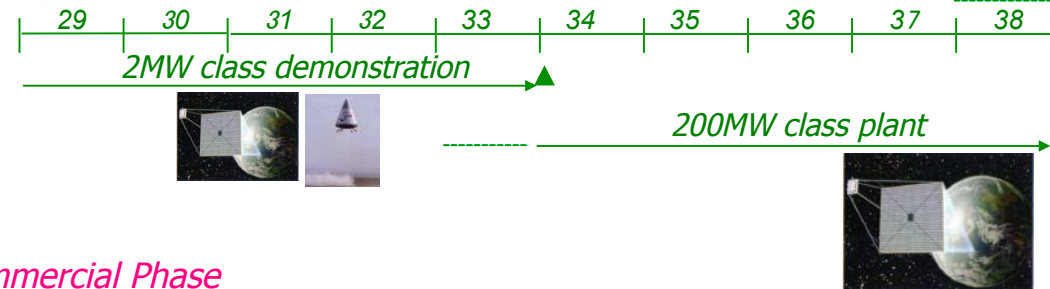
# Roadmap towards SPS Commercialization

If we follow the roadmap below, we will be able to **realize the first commercial Solar Power Satellite of 1GW class in mid-2040's**. Increasing the number of the Solar Power Satellites year by year, both the issues, the depletion of energy resources and deterioration of the global environment, will be overcome before 2100.

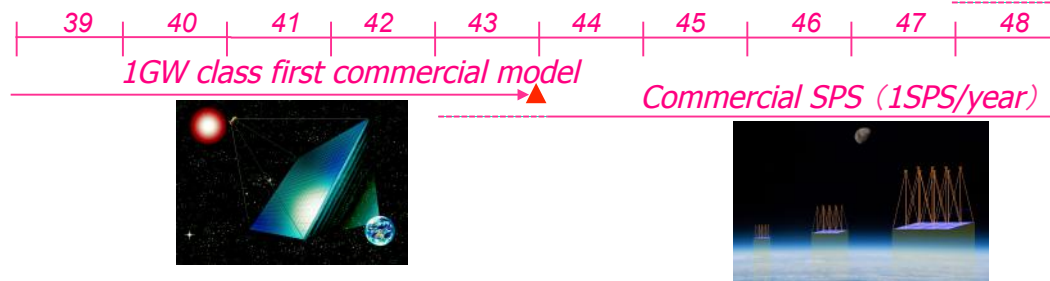
## Research Phase



## Development Phase



## Commercial Phase





# SPS Requirements for Space Transportation

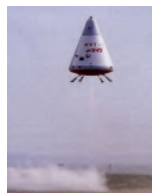
Phase	Small scale demonstration	Large scale demonstration	Small plant	Large plant	First commercial model	Commercial
Target year	~2015	~2020	~2025	~2030	~2035	2035~
Orbit	LEO	LEO	1000 km	GEO	GEO	GEO
Power level	1~5kW	100 kW	2 MW	200 MW	1 GW	1 GW
System weight	500 kg	15 tons	42.5 tons	5300 tons	26600 tons	26600 tons
Construction	NA	NA	6 months	3 years	5 years	1 year
Payload weight	500 kg	15 tons	10 tons	50 tons	50 tons	50 tons
Launch vehicle	Small ELV LEO	Large ELV LEO	1 RLV 1000km 5 Round trips 1 launch/month	1 RLV 500 km 207 round trips 1 launch/5 days 5 days turn around	3 RLV 500 km 345 round trips 1 launch/2 days 5 days turn around	15 RLV 500 km 69 round trips 2.8 launch per day 5 days turn around
Orbit transfer vehicle	NA	NA	NA	14 OTV 500 km-GEO 9 round trips*	42 OTV 500 km-GEO 15 round trips*	206 OTV 500 km-GEO 3 round trips*



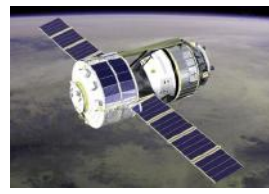
Epsilon Launch Vehicle



H-II Launch Vehicle



Reusable Sounding Rocket



OTV (Image)

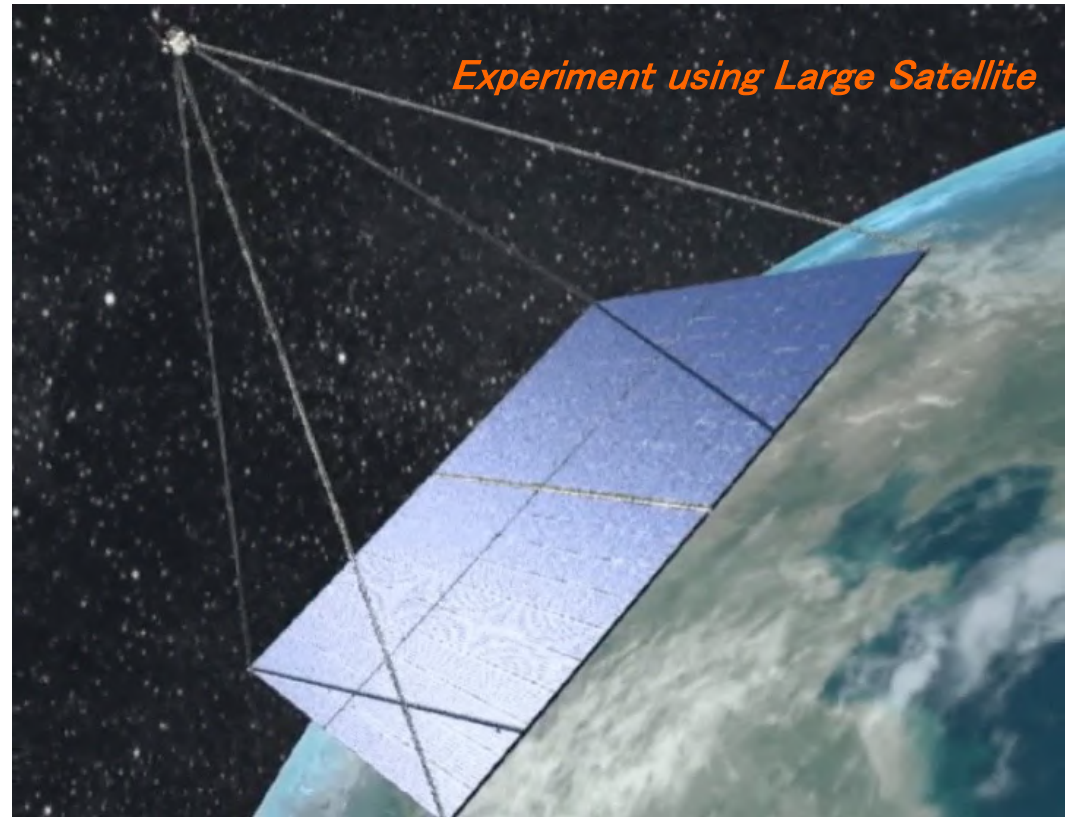
\* : fuel 38 tons for a round trip

*Space experiment for demonstrating wireless power transmission technology that should be carried out at an early stage*



*Experiment using small satellite*



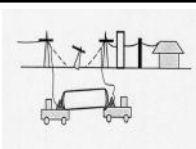




*Altitude: about 400km  
Power level: about 1–5kW.  
Demonstration of microwave beam control technologies and propagation in the ionosphere.*



*Experiment using Large Satellite*

*Altitude: about 400km  
Power level: about 100–400kW  
Demonstration of power transmission.*

# *Spin-off of wireless power transmission technologies (application to industrial and space development)*

<i>Usage</i>	<i>Example</i>	<i>Illustration, photos</i>	<i>Required microwave technology</i>
<i>Power supply to maintenance, inspection, and diagnostic equipment in difficult situations for power line.</i>	<i>Power supply to forest environment survey sensor network</i>		<i>Transmission technology Receiving technology Direction control (coarse)</i>
<i>Small power supply for disasters (weak power supply from balloons, etc.)</i>	<i>Transmission of power and information from airships (Kyoto University, 2009)</i>		<i>Transmission technology Receiving technology Direction control (coarse)</i>
<i>Medium-sized power transmission (in disaster, temporary use)</i>	<i>Wireless connection of disconnected transmission lines (Takano, 2011)</i>		<i>Transmission technology Receiving technology Direction control (fine)</i>
<i>Microwave power space aimed at supplying power to electronic devices (low power density)</i>	<i>Wireless power space (Kyoto Univ.), Wireless charging (Ossia demo.)</i>		<i>Transmission technology Receiving technology Direction control (coarse)</i>
<i>Power supply to batteries of vehicles, etc.</i>	<i>Charging a car battery (Mitsubishi Heavy Industries, Kyoto University)</i>		<i>Transmission technology Receiving technology</i>
<i>Power supply to flying objects</i>	<i>Power transmission to airships, small planes, and small helicopters</i>		<i>Transmission technology Receiving technology Direction control (fine)</i>
<i>Wireless power supply system for lunar/planetary exploration</i>	<i>Lunar microwave power transmission (University of Colorado)</i>		<i>Transmission technology Receiving technology Direction control (fine)</i>

## ***8. Concluding remarks***



# *If SPS system is realized ...*



- limitless clean energy will be obtained,*
- global environment will be restored and preserved,*
- international conflicts over energy resources will be terminated,*
- creative and comfortable society will be established,*
- ▪ ▪ and, new civilization and culture will be developed in the creative and vigorous society expanding into space.*